# of Transportation

## **MAJOR REPAIR AND ALTERATION** (Airframe, Powerplant, Propeller, or Appliance)

	Form	Approv
- 1		

OMB No. 2120-0020

For FAA Use Only Office Identification

Federal	Aviation
Adminic	tration

FAR 43 9 FAR 43 A

and disposition	of this form. This	report is required l	by law (4	9 U	.S.C. 1421). Fai						
for each violatio	on (Section 901 of	Federal Aviation	Act of 19	58).	<u> </u>	I Marta	<u> </u>		<u> </u>	2 4	
	Make	David				Model					
1. Aircraft	Serial No.	Ryan	K V			Navion B  Nationality and Registration Mark					
I. All Clait	Seliai IVO.	NAV 12	212D								
	Name (As shown o	NAV-4-2313B  Name (As shown on registration certificate)				Address (As shown on registra			N5413K		
	Name (As shown s	Putney, Willia		П					almoral Drive		
2. Owner						Oakland, CA 94619					
		itougoio, c	Outstand, Cri 74017								
				3. F	or FAA Use Or	nly	Year of the second	1 11 17			
1 1 6			n, mili Ngariwi ka Tiliki kale	0	5-29-03 DATE	SIGNATU	ith the applicable airwove described aircraft, strized in FAR 43, Section			a y kituli 11 y kituli 11 kirikan	
		/ 1	Est, Carl	4. Unit Identification			Section of the section	i dazada	5. Type		
Unit	M	Make			Model	22.5	Serial No.		, Repair	Alteration	
AIRFRAME		~~~~~ (As described in Item 1 above) ~~~~~~							X		
POWERPLANT						on in ac			todomike	029E (S	
PROPELLER		in an survey of 1 (its ext equests			teriore in company of a company. Note: Section of the company of t				*12	2000 ) (B. 2000	
APPLIANCE	Туре		J						para log gif	rend (A	
	Manufacturer	Manufacturer									
			6.	Cor	nformity Stater	nent				(C) (C) (C)	
A. Agency's Nar	me and Address				Kind of Agency			C. Cert	ificate No.		
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	875A Island D	r. #253		7E 7	Foreign Certificated Mechanic			A.P. 2020552 I.A.			
	Alameda, CA.		1 1 2 2 4 1	Certified Repair Station			- 20203321.71.				
Maineda, CA. 74502				Manufacturer							
hereto have	t the repair and/or e been made in ac furnished herein i	ccordance with the	e requirer	men	its of Part 43 of	the U.S. Fe					
6-2-03				Signature of Authorized Individual ABorduoe							
<u> </u>			7 App		al for Return To	Comico	1100	- Comment	_	*********	
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	authority given pe the Federal Aviat					m 4 was ins		anner pre	escribed by th	e	
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BY FAA [	Designee	Repair Station		Tra	erson Approved ansport Canada rworthiness Gro		to kiriki, ostannik sijak	1 48 11 9 21 11 2	ti i kangasan		
Date of Approval or Rejection  6-2-03  Certificate or Designation No.  A.P. 2020552 I.A.					Signature of Authorized Individual ABorduos						

#### NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

### 8. Description of Work Accomplished

(If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

**Description of alteration:** This alteration removes a Narco Mk-12A NAV/COM, CDI, VOR Resolver and Power Unit. This alteration also installs a TKM, Inc model MX-12 NAV/COM and a TKM model MC60 Navigation Converter and Indicator.

Description of work: The Narco radio and its wiring and accessories were removed and discarded. Installation of the new radio and indicator were accomplished according to the installation instructions in the TKM "MX12 Nav-Comm Owners Manual" and the TKM "MC60 Navigation Converter and Indicator Operation and Installation" manual. The MX-12 and the MC60 are located in the instrument panel at 7a and 6a respectively (see attached drawing N5413K Panel). Audio inputs and outputs are provided to the aircraft's audio panel (see attached N5413K Audio drawing). The existing communication and navigation antennas were reused for this installation.

A dedicated 10A circuit breaker (Klixon PN: 7277-2-10) circuit breaker located in the switch panel to the left of the pilot labeled "Nav/Comm 2" provides power to the radio and CDI. The total aircraft system electrical load does not exceed 80% of the generating capacity after this alteration. Connections to aircraft power comply with 23.1365 "Electrical cables and equipment" (a, d, e) and 23.1357 "Circuit protective devices". The systems conform to 23.1311 "Electronic display instrument systems" (a)(1,2,6).

After installation a check for interactions with other systems was made to comply with 23.1309 "Equipment, systems and installations" (a)(1).

A new weight and balance measurement in accordance with 43.13 chapter 10 has been done which includes this alteration.

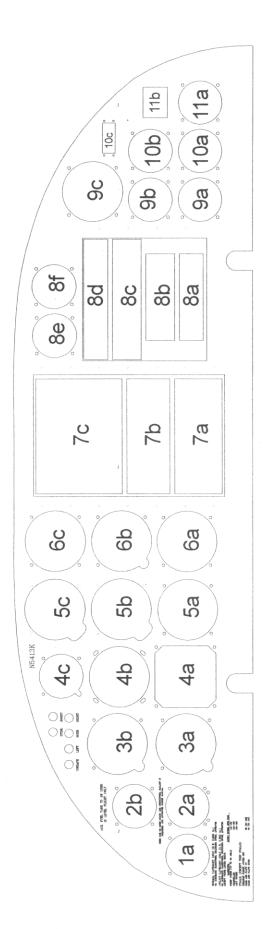
#### INSTRUCTIONS FOR CONTINUED AIRWORTHINESS

- 1) Introduction: See above (Form 337 section 8).
- 2) Description: See above (Form 337 section 8).
- 3) Control: Complete operational information is included in the TKM "MX12 Nav-Comm Owners Manual" and the TKM "MC60 Navigation Converter and Indicator Operation and Installation" manual.
- 4) Servicing information: Not applicable.
- 5) Maintenance Instructions: See TKM "MX12 Nav-Comm Owners Manual"
- 6) Trouble shooting information: See TKM "MX12 Nav-Comm Owners Manual".
- 7) Removal and replacement information: The MX12 can be removed from its tray by turning the ¼ turn fastener on the rear of the mounting tray counterclockwise. Disconnect the multi-pin connectors and the BNC connectors that connect to the aircraft wiring. When free MX12 can be pulled out. The MC60 is attached to the instrument panel by 4-#6-32 flat head screws. Disconnect the multi-pin connector on the rear of the unit. If the aircraft is to be returned to service without this unit installed, insure that cables and connectors are secured out of the way of flight controls.
- 8) Diagrams: Not applicable.
- 9) Special inspection requirements: Not applicable.
- 10) Application of protective treatments: Not applicable.
- 11) Data: 8 #6-32 flat head screws were used to attach the MX-12's tray to the radio rack.
- 12) List of special tools: No special tools are required to install or maintain any components associated with this alteration.
- 13) For commuter category aircraft: Not applicable.
- 14) Recommended overhaul periods: Not applicable.
- 15) Airworthiness Limitation Section: Not applicable.

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**16) Revision:** A letter will be submitted to the local FSDO with a copy of the revised FAA Form 337 and revised ICA. The FAA inspector accepts the change by signing Block 3 of the 337.

x Additional Sheets are Attached



Material: 6061-T6

Thickness: 0.100"

Finish: Low reflectivity powder coating

SN: NAV-4-2313B | N5413K | Pane | Reg: N5413K

SIZE FSCM NO. NAV-2452781-13888 1.0 SCALE 1:5 DATE 22 May, 2003 SHEET 1 Of 1

