of Transportation

MAJOR REPAIR AND ALTERATION (Airframe, Powerplant, Propeller, or Appliance)

	Form	Approve
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OMB No. 2120-0020

For FAA Use Only

Federal Aviation Administration

Office Identification INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions

for	dispo each v	osition of violation	this form. This (Section 901 of	report is required by Federal Aviation Ac	law t of 1	(49 l 1958)	J.S.().	C. 1421). Failure t	to report	can	result in a civil p	enalty no	t to exceed \$1	,000
			Make			,	,-		Model		-			
			Ryan]	Navion 1	В	
1. Aircraft		ft	Serial No.						National	ity and	Registration Mark			
_			NAV-4-2313B					* ÷	11			N5413k		
			Name (As shown	on registration certificate					Address	(As s	hown on registratio		,	
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	Un	IIL ·		Make				Model			Serial No.		Repair	Alteration
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POV	VERP	PLANT												
PRO	PELL	LER												
			Туре											
APP	LIAN	CE												
			Manufacturer											
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Alameda, CA. 9450			. 94502				Certified Repair S	Station						
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BY		FAA De		Repair Station				rson Approved by nada Airworthines						
Date		_	Rejection	Certificate or Designation No.			Sig	nature of Authoriz	zed Indiv	idual	PBo1	dus	e	
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NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. [Description of Work Accomplished	al sheets. Identify with aircraft nationality and registration mark and date work completed.)
	(п тюге space is required, апасл адашота	See Attached ICA.
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		Additional Sheets are Attached

<u>INSTRUCTIONS FOR CONTINUED AIRWORTHINESS</u> INSTRUCTIONS FOR CONTINUED AIRWORTHINESS

A/C Make: Ryan	Model: Navion B	S/N: NAV-4-2313B	Reg. #: N5413K
Revision:	Date:		
This sixteen item checklist are Instru Airworthiness (HBAW 98-18 Dated O installed:	october 7, 1998), are applications	worthiness (ICA), to comply wable to the aircraft listed above	with FAA Handbook Bulletin for when the following equipment is
SYSTEM:			

ITEM	CHECKLIST INFORMATION
1.	Introduction: This section briefly describes the aircraft, engine, propeller, or component that has been altered. Include any other information on the content, scope, purpose, arrangement, applicability, definitions, abbreviations, precaution units of measurement, referenced publications, and distribution of the ICA as applicable.
<i>y</i> - 1	Comment: This alteration is to replace the main fuel tank fuel level sender and fuel level gauge. This was necessitated by the failure of the original main fuel tank fuel level sender and the unavailability of a direct replacement part. The King-Seeley company manufactured the original sender and gauge. King-Seeley has been out of business since the mid 1950's. A search for a certified repair station or a replacement sender found none available. Therefore a project to develop a replacement sender was undertaken by the owner with the concurrence of the Administrator. In addition to the new sender a new fuel level gauge will be installed. The negauge is an aircraft type gauge, manufactured by Electronics International of Bend, OR, and currently being used in many certificated aircraft as an alteration.
2.	Description: Of the major alteration, its functions, including an explanation of its interface with other systems, if any.
	Comment: No alteration to the airframe, fuel system or electrical system was made as a part of this alteration except as described herein.
	A part number MC0726110-1 replacement sender, made under PMA by Mc Farlane Aviation, was used as the component basis for the fabrication of the fuel level sender. The MC0726110-1 sender uses a standard Stewart-Warner style mounting and would not mount to the aircraft fuel tank. The sender was disassembled and a new mounting plate designed. Design of the new mounting plate involved determining the required sender float arm and tank geometry and then incorporating this information into the new sender mounting plate in such a manner that the new sender measures the greatest possible volume of the fuel tank (see attached drawings). A certified shop, using the same type of material used in the Mc Farlane sender's mounting plate, fabricated the mounting plate. The float arm was made from materials obtained from Mc Farlane Aviation with the sender.
	Because the King-Seeley and Stewart-Warner senders are of fundamentally different electrical designs the fuel level gauge was also replaced. The Electronics International FL-2RA-12 gauge was chosen because it can be calibrated to accurately display the fuel level with the new sender. The gauge will also replace the existing auxiliary tank fuel level gauge. The auxiliary fuel tank already had a Stewart-Warner type sender and no other alteration was necessary to that system. Both the Stewart-Warner and the King-Seeley systems use a single wire between the sender and the gauge so no aircraft wiring changes were required.
	Both the main and auxiliary tanks were calibrated according to Electronic International's calibration procedure. The main fuel tank calibration followed the procedure outlined in the Ryan Navion 1951 Service manual with respect to measured volumes and tank pre-fill prior to indication above empty.
3.	Control: Operation information: Or special procedures, if any.
1	Comment: No special procedures are required. Servicing information: Such as types of fluids used, servicing points, and location of access panels, as appropriate.
4.	Comment: Not applicable.
5.	Maintenance Instructions: Such as recommended inspection/maintenance periods in which each of the major alteration components are inspected, cleaned, lubricated, adjusted, tested, including applicable wear tolerances and

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	work recommended at each scheduled maintenance period. This section can refer to the manufacturers' instructions for the equipment installed where appropriate (e.g., functional checks, repairs, inspections.) It should also include any special notes, cautions, or warnings, as applicable.
	Comment: No continuing maintenance required.
6.	Trouble shooting information: Information describing probable malfunctions, how to recognize those malfunctions, and the remedial actions to be taken.
	Comment: Follow trouble shooting information contained in the EI FL-2 installation and owners manual.

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7.	Removal and replacement information: This section describes the order and method of removing and replacing products, parts and any necessary precautions. This section should also describe or refer to manufacturer's instructions to make required tests, trim checks, alignment, calibrations, center of gravity changes, lifting or shoring, etc., if any.
•	Comment: The battery master switch(s) must be in the "Off" position before removal of the gauge or sender for maintenance.
8.	Diagrams: Of access plates and information, if needed, to gain access for inspection.
	Comment: Not applicable.
9.	Special inspection requirements: Such as X-ray, ultrasonic testing, or magnetic particle inspection, if required.
	Comment: Not applicable
10.	Application of protective treatments: To the affected area after inspection and/or maintenance, if any.
	Comment: Not applicable.
11.	Data: Relative to structural fasteners such as type, torque, and installation requirements, if any.
	Comment: There are no structurally significant fasteners associated with this alteration.
12.	List of special tools: Special tools that are required, if any.
	Comment: No special tools are required to install or maintain any components associated with this alteration.
13.	For commuter category aircraft: The following additional information must be furnished, as applicable:
	A. Electrical loads B. Methods of balancing flight controls
	C. Identification of primary and secondary structures
	D. Special repair methods applicable to the airplane.
	Comment: Not applicable.
14.	Recommended overhaul periods: Are required to be noted on the ICA when an overhaul period has been set by the manufacturer of a component, or equipment. If there is no overhaul period, the ICA should state for item 14: "No additional overhaul time limitations."
	Comment: No component associated with this alteration has an overhaul period.

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15.	Airworthiness Limitation Section: Include any "approved" airworthiness limitations identified by the manufacturer or FAA Type Certificate Holding Office (e.g., An STC incorporated in a larger field approved major alteration may have an airworthiness limitation.) The FAA inspector should not establish, alter, or cancel airworthiness limitations without coordinating with the appropriate FAA Type Certificate Holding Office. If there are no changes to the airworthiness limitations, the ICA should state for item 15: "No additional airworthiness limitations" or "Not Applicable." Comment: Not applicable.
16.	Revision: This section should include information on how to revise the ICA. For example, a letter will be submitted to the local FSDO with a copy of the revised FAA Form 337 and revised ICA. The FAA inspector accepts the change by signing Block 3 and including the following statement: "The attached revised/new Instructions for Continued Airworthiness (date) for the above aircraft or component major alteration have been accepted by the FAA, superseding the Instructions for Continued Airworthiness (date)." Once the revision has been accepted, a maintenance record entry will be made, identifying the revision, its location, and date of the Form 337. Comment: a letter will be submitted to the local FSDO with a copy of the revised FAA Form 337 and revised ICA. The FAA inspector accepts the change by signing Block 3 of the 337.

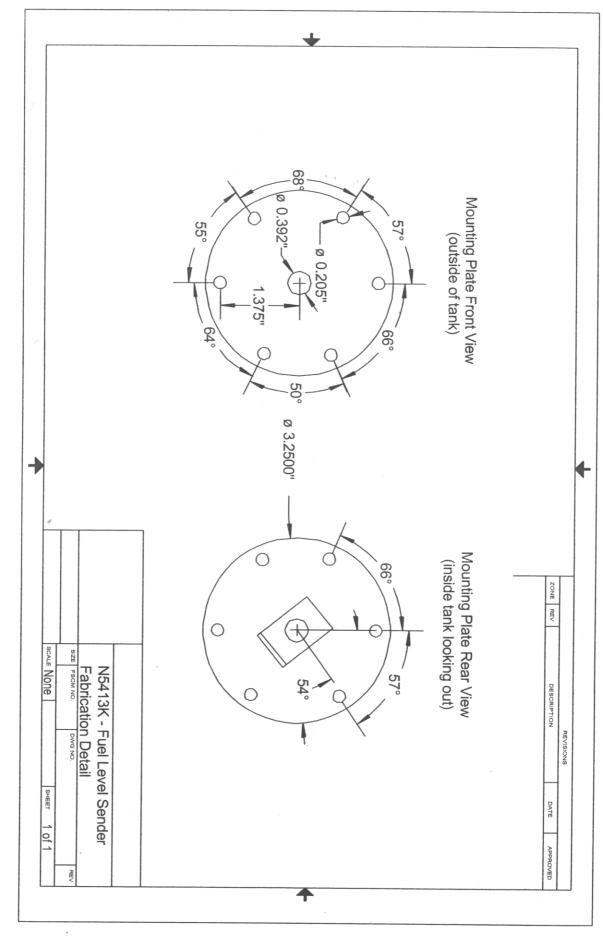
Note:

Implementation and Record Keeping: For major alterations performed in accordance with FAA Field Approval policy, the owner/operator operating under part 91 is responsible for ensuring that the ICA is made part of the applicable section 91.409 inspection program for their aircraft. This is accomplished when a maintenance entry is made in the aircraft's maintenance record in accordance with section 43.9. This entry records the major alteration and identifies the original ICA location (e.g., Block 8 of FAA Form 337, dated 5/28/98) along with a statement that the ICA is now part of the aircraft's inspection/maintenance requirements.

For major alterations performed in accordance with a field approval on air carrier aircraft, the air carrier operator is responsible for ensuring that the ICA is made part of the applicable inspection/maintenance program for their aircraft. If a procedure is not currently included in the operator's manual to incorporate ICA, this process will need to be appropriately addressed (i.e. the operator submits a revision to its maintenance program to the applicable certificate-holding district office (CHDO).

For aircraft inspected under an Approved Aircraft Inspection Program (AAIP), the operator will submit a change to the CHDO in accordance with section 135.419 b).

For air carrier aircraft inspected using an annual/100 hour inspection program, a reference to the new ICA will be made in the aircraft's maintenance record in accordance with section 43.9. This entry records the major alteration and identifies the original ICA location (e.g., ICA are located/attached to Block 8 of FAA Form 337, dated 5/28/98). In addition, the operator will request a revision to the operator's Operations Specifications, additional maintenance requirements, which incorporates the ICA into the inspection program.



REG. No. . . . K SN: NAV-4-2313B

